IMU Noise and Characterization

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1. Motivation

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Figure: From Gyro Measurements to Orientation

Motivation for Modelling IMU Noise: Example

Gyro Integration: nonlinear motion, noise, bias

Figure: Error from integrating Gyro Measurements without dealing with noise

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Motivation for Modelling IMU Noise: IMU Noise **Components**

Fig. 1. Inertial sensor model with a deterministic and a random component. Here, the true angular rates ω are corrupted with deterministic errors, for example a scale factor that varies with temperature, as well as nondeterministic errors, such as additive broadband noise. This report presents a method to identify noise processes according to their contribution to the angular increments Ω .

Figure: IMU Noise Compon[en](#page-4-0)t[s](#page-6-0)

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Motivation for Modelling IMU Noise: Types of IMU Noise

- Quantization Noise
- Angle / Velocity Random Walk Noise
- **Correlated Noise**
- **Bias Instability Noise**
- Rate / Acceleration Random Walk Noise

2. Power Spectral Density

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Fourier Transform

Figure: Fourier Transform

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- Power: refers to the fact that PSD is the mean-square value of the signal being analyzed
- **Spectral**: refers to the fact PSD is a function of frequency, it represents distribution of a signal over a spectrum of frequencies
- **Density:** refers to the fact that the magnitude of PSD is normalized to a single hertz bandwidth.

If the signal being analyzed is a Wide-Sense Stationarity (WSS) discrete random process, according to the Wiener-Khinchin theorem the PSD is defined as:

$$
P(f) = \sum_{m=-\infty}^{\infty} R_{xx}(m) \exp(-j2\pi fm)
$$
 (1)

Where $R_{xx}(f)$ is the **Autocorrelation function** of the random process $X(t)$ and τ is the time lag:

$$
R_{xx}(f) = E[X(t)X(t-\tau)] \tag{2}
$$

- A Random Process is Stationary if its statistical properties do not change in time
- WSS is also known as Weak-Sense Stationarity, Covariance Stationarity or Second-Order Stationarity.
- The main thing to know is a random process is WSS if its **mean** and its correlation function do not change by shifts in time.

Autocorrelation Function

Autocorrelation is the degree of similarity between a given time series and a lagged version of itself over successive time intervals

$$
R_{xx}(f) = E[X(t)X(t-\tau)] \tag{3}
$$

Figure: Autocorrelation in Action

- In most practical situations, the PSD of a random process is not available.
- Can estimate a given signal's power spectral density by taking magnitude squared of its Fourier transform as the estimate of the PSD

One form is:

$$
P(f_k) = \frac{1}{N} \left| \sum_{n=0}^{N-1} x_n \exp\left(-\frac{j2\pi k n}{N}\right) \right|^2 \tag{4}
$$

If we compare DFT

$$
|X(f_k)| = \left|\sum_{n=0}^{N-1} x[n] \exp\left(-j2\pi nk/N\right)\right|
$$

And PSD (estimate)

$$
P(f_k) = \frac{1}{N} \left| \sum_{n=0}^{N-1} x[n] \exp(-j2\pi f_k n) \right|^2 \tag{6}
$$

You have:

$$
P(f_k) = \frac{1}{N} |X(f_k)|^2 \tag{7}
$$

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Power Spectral Density: Color of Noise

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In summary:

- \bullet DFT \neq PSD
- DFT: shows the spectral content of the signal (amplitude and phase of harmonics)
- **PSD**: describes how the power of the signal is distributed over frequency by performing the mean-square on the signal value.

3. Allan Variance

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Also called a Two-Sample Deviation, or square-root of the Allan Variance, where:

$$
\sigma_{\mathbf{y}}^2(\tau) = \frac{1}{2} \langle (\bar{y}_{n+1} - \bar{y}_n)^2 \rangle \tag{8}
$$

$$
=\frac{1}{2\tau^2}\langle (y_{n+2}-2y_{n+1}+y_n)^2\rangle \tag{9}
$$

Where τ is the observation period, \bar{v}_n is the n-th fractional frequency average over the observation time τ . The samples are taken with no dead-time between them, which is achieved by letting time period $T = \tau$.

- 1. Acquire time series data on gyroscope or accelerometer
- 2. Set average time to be $\tau = m\tau_0$, where m is the averaging factor. The value of m where $m < (N-1)/2$.
- 3. Divide time history of signal into clusters of finite time duration of $\tau = m\tau_0$

- 4. Once clusters are form, compute the Allan Variance
	- Calculate θ corresponding to each gyro output sample, this can be accomplished as in.

$$
\theta(t) = \int^t \Omega(t')dt' \tag{10}
$$

• Once N values of θ have been computed, calculate the Allan Variance σ^2 represents as a function of τ where $\langle \cdot \rangle$ is the ensemble average.

$$
\sigma^2 = \frac{1}{2\tau^2} \left\langle \left(\theta_{k+2m} - 2\theta_{k+m} + \theta_k\right)^2 \right\rangle \tag{11}
$$

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5. Calculate Allan Deviation (AD) value for a particular τ . This can be obtained simply by square rooting the Allan Variance (AVAR). This result will now be used to characterize the noise in a gyroscope.

$$
AD(\tau) = \sqrt{AVAR(\tau)} \tag{12}
$$

Figure: Characteristics of an Allan Deviation Plot (For Gyroscope)

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Figure: Gyrometer Noise Characterization

Figure: Accelerometer Noise Characterization

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4. IMU Noise Model

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The standard noise model:

$$
z = x + v \tag{13}
$$

$$
\dot{x} = \frac{1}{\tau_b} x + \omega \tag{14}
$$

Where:

- z is the modelled noise process
- \bullet x is the slowly varying process with correlation time τ_b , "driven" by another independent white noise w.
- \bullet v is the white noise component

IMU Noise Model: Discrete version

Algorithm 1 Discrete-Time Equivalent of the Standard Noise Model (1a)

init: $\sigma_{wd}^2 \leftarrow \frac{1}{\Delta t} \sigma_w^2$ \rightarrow assuming v is band-limited to $\frac{1}{2\Delta t}$
 $\sigma_{kd}^2 \leftarrow \Delta t \sigma_k^2$ \rightarrow assuming $\tau_b \gg \Delta t$ \triangleright assuming $\tau_b \gg \Delta t$ $\Phi_d \leftarrow \exp\left(-\frac{1}{\tau_b}\Delta t\right)$ $x_0 \leftarrow \begin{cases} 0 & \text{if } \frac{1}{\tau_b} = 0 \text{ (by definition)} \\ \mathcal{N}\left(0, \frac{\sigma_b^2 \tau_b}{2}\right) & \text{otherwise} \end{cases}$

for $k \leftarrow 1$ to n do $w_k \leftarrow \mathcal{N}\left(0, \sigma_{bd}^2\right), v_k \leftarrow \mathcal{N}\left(0, \sigma_{wd}^2\right)$ $x_k \leftarrow \Phi_d x_{k-1} + w_k$ $z_k \leftarrow x_k + v_k$ end for

Figure: IMU Noise Model: Discrete version

5. IMU Pre-Integration

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Realtime is difficult as map and trajectory grows overtime, there are generally 3 approaches towards realtime operation:

- PTAM
- Marginalization (fixed-lag smoothing)
- **•** Filtering

But PTAM has a keyframe limit, filtering and marginalization commit to a linearization point when marginalizing which introduces drift and potential inconsistencies.

IMU Pre-Integration: Bundle Adjustment (structured)

Figure: Bundle Adjustment

IMU Pre-Integration: Bundle Adjustment (structured)

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IMU Pre-Integration: Approach

Fig. 4: Left: visual and inertial measurements in VIN. Right: factor graph in which several IMU measurements are summarized in a single preintegrated IMU factor and a structureless vision factor constraints keyframes observing the same landmark.

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- Avoid repeated integration by defining relative motion increments
- **Assume bias** is known and constant
- Make Bundle Adjustment problem structureless by "Lifting" the cost function

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IMU Pre-Integration: Results

Fig. 6: Left: two images from the indoor trajectory dataset with tracked features in green. Right: top view of the trajectory estimate produced by our approach (blue) and 3D landmarks triangulated from the trajectory (green).

Figure: IMU Pre-integration Results

IMU Pre-Integration: Results

Fig. 7: Comparison of the proposed approach versus the ASLAM algorithm <a>[9] and an implementation of the MSCKF filter<a>[20]. Relative errors are measured over different segments of the trajectory, of length $\{10, 40, 90, 160, 250, 360\}$ m, according to the odometric error metric in [46].

Figure: IMU Pre-integration Results

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Questions?

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1. What are the definitions of these terms?

- Quantization Noise
- Angle / Velocity Random Walk Noise
- **Correlated Noise**
- Bias Instability Noise
- Rate / Acceleration Random Walk Noise
- 2. Simulate an IMU using the standard noise model
- 3. Plot Fourier Transform and Power Spectral Density of simulated IMU
- 4. Extract the IMU Noise characteristics using Allan Variance

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